

Conservation of Energy (Contd.....)

Yesterday we saw the expression for conservation of energy using RTT.

$$\frac{dQ}{dt} - \frac{dW_s}{dt} - \frac{dW_p}{dt} = \frac{\partial}{\partial t} \left[\iiint_{CV} \left(\hat{u} + \frac{v^2}{2} + gz \right) \rho dU \right] + \iint_{CS} \left(\hat{h} + \frac{v^2}{2} + gz \right) \rho (\vec{v} \cdot \hat{n}) dA$$

(Of course, the above expression is for stationary control volume)

Subsequently for one dimensional inlet & outlet having steady fluid flow, the energy equation becomes

$$\frac{dQ}{dt} - \dot{W}_s - \dot{W}_p = -\dot{m}_{in} \left[\hat{h}_{in} + \frac{v_{in}^2}{2} + gz_{in} \right] + \dot{m}_{out} \left[\hat{h}_{out} + \frac{v_{out}^2}{2} + gz_{out} \right]$$

As for steady flow, $\dot{m}_{in} = \dot{m}_{out} = \dot{m}$

$$q = \frac{\frac{dQ}{dt}}{\dot{m}} = \frac{\dot{Q}}{\dot{m}} = \frac{dQ}{dm}$$

$$w_s = \frac{\dot{W}_s}{\dot{m}} = \frac{dW_s}{dm}$$

$$w_p = \frac{\dot{W}_p}{\dot{m}} = \frac{dW_p}{dm}$$

$$\hat{h}_1 + \frac{v_1^2}{2} + gz_1 = \left(\hat{h}_2 + \frac{v_2^2}{2} + gz_2 \right) - q - w_s + w_p$$

Quick Example (Adopted from FM White's fluid Mechanics)

Air with gas constant $R = 287 \text{ J/kg-K}$ & specific heat C_p , flows steadily as shown, through a turbine that produces 521.85 kJ of energy. For the inlet & exit conditions shown in figure, estimate a) the velocity at the outlet V_2 b) the heat transfer Q

**Solution:**

Given, $D_1 = 15 \text{ cm}$, $T_1 = 423 \text{ K}$, $p_1 = 1.034 \times 10^6 \text{ Pa}$, $V_1 = 30 \text{ m/sec}$;

$D_2 = 15 \text{ cm}$, $T_2 = 275 \text{ K}$, $p_2 = 2.758 \times 10^5 \text{ Pa}$, $V_2 = ?$

Also given, $R = 287 \text{ J/kg-K}$; $C_p = 1.005 \text{ kJ/kg-K} = 1005 \text{ J/kg-K}$

As the flow is steady & one dimensional,

From perfect gas law: $p = \rho RT$

where, p = gas pressure (P_a)

ρ = density of gas

R = universal gas constant

T = temperature in Kelvin

$$p_1 = \rho_1 R T_1$$

&

$$p_2 = \rho_2 R T_2$$

$$\rho_1 = \frac{p_1}{R T_1} = \frac{1.034 \times 10^6}{287 \times 423} = 8.52 \text{ Kg/m}^3$$

$$\rho_2 = \frac{p_2}{R T_2} = \frac{2.758 \times 10^5}{287 \times 275} = 3.50 \text{ Kg/m}^3$$

Mass flow rate, $\dot{m} = \dot{m}_1 = \dot{m}_2$ (as, flow is steady)

$$\dot{m}_1 = \rho_1 A_1 v_1 = 8.52 * \frac{\pi}{4} * (0.15)^2 * 30 = 4.514 \text{ Kg/s}$$

$$\dot{m}_1 = \dot{m}_2 = \rho_2 A_2 v_2 = 3.50 * \frac{\pi}{4} * (0.15)^2 * v_2$$

$$4.514 = 3.50 * \frac{\pi}{4} * (0.15)^2 * v_2$$

$$v_2 = 73 \text{ m/s}$$

As, the flow is steady, the energy equation from RTT

$$\frac{dQ}{dt} - \frac{dW_s}{dt} = \dot{m} \left[\hat{h}_2 + \frac{v_2^2}{2} + gz_2 \right] - \dot{m} \left[\hat{h}_1 + \frac{v_1^2}{2} + gz_1 \right]$$

Given the turbine produces 700 horsepower = 514.5 kJ/sec

$$\text{So, } \frac{dQ}{dt} - 514500 = 4.514 \left[1005 * 275 + \frac{73^2}{2} + 0 \right] - 4.514 \left[1005 * 423 + \frac{30^2}{2} \right] = -150 \text{ kW}$$

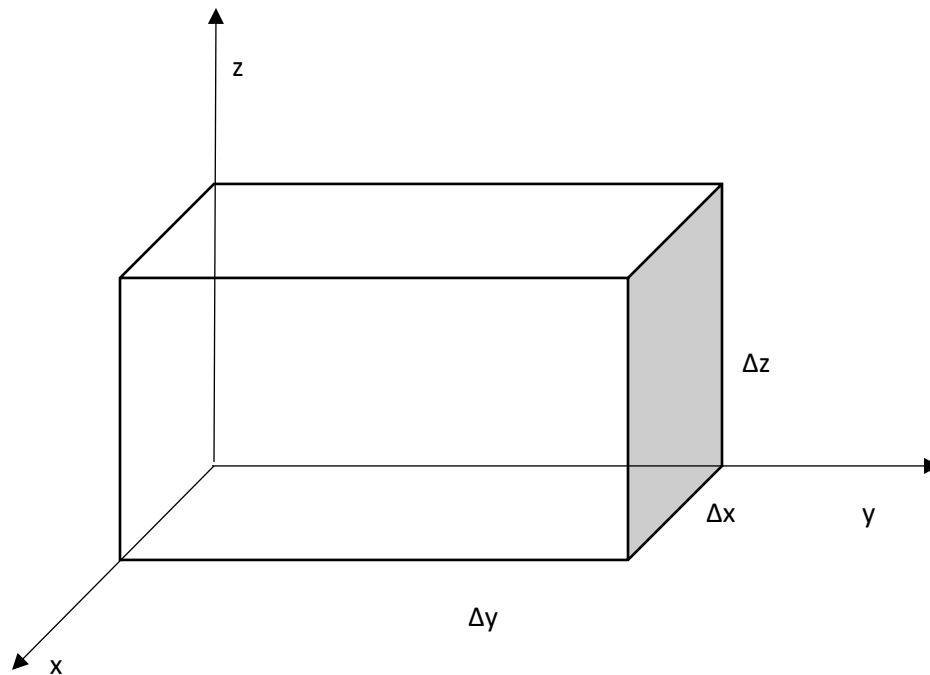
The potential quantity is cancelled as z_1 & z_2 are at same elevation.

DIFFERENTIAL RELATIONS FOR FLUID FLOW

- Till now we were discussing on the integral or control volume relations for fluid flow.
- We used Reynold's Transport Theorem for the control volume analysis.
- We have also seen through several examples, the control volumes were of different sizes according to the problem concerned.
- In fluid mechanics, to analyze fluid flow, one can also adopt analysis that involve infinitesimally small volume rather than the large control volume.
- This infinitesimally small volume may be associated with mathematical point in spaces.
- Today, we will start discussing about the Differential Approach or Infinitesimal Approach.
- We will be using the Reynold's Transport Theorem for an infinitesimally small control volume.

Differential Equation of Mass Conservation

Let us consider a small rectangular prism shaped element of fluid that is flowing.



As seen in the figure, the size of the element is given by $\Delta x \Delta y \Delta z$ & corresponding coordinate axis is also suggested.

Let velocity vector be $\vec{v} = u\hat{i} + v\hat{j} + w\hat{k}$

In the y direction, let the mass influx on the left side be = Density×Area ×Velocity

$$= \rho v \Delta x \Delta z$$

And the outflux in y direction

$$= [\rho v + \frac{\partial(\rho v)}{\partial y} \Delta y] \Delta x \Delta z$$

Similarly, we assign mass fluxes in x & z direction also,

| Direction | Mass Influx | Mass Outflux |
|-----------|----------------------------|---|
| x | $\rho u \Delta y \Delta z$ | $[\rho u + \frac{\partial(\rho u)}{\partial x} \Delta x] \Delta y \Delta z$ |
| z | $\rho w \Delta x \Delta y$ | $[\rho w + \frac{\partial(\rho w)}{\partial z} \Delta z] \Delta x \Delta y$ |

Applying the Reynold's Transport Theorem (conservation of mass) on this rectangular prism for fluid

$$\left. \frac{dB}{dt} \right|_{system} = \frac{Dm}{Dt} = 0 = \frac{\partial}{\partial t} \iiint_{CV} \rho dU + \iint_{CS} \rho(\vec{v} \cdot \hat{n}) dA$$

Since one dimensional inputs & outputs are present, & the rectangular volume is not shrinking or expanding;

$$0 = \iiint_{cv} \frac{\partial \rho}{\partial t} dU + \sum [\rho A v]_{outlet} - \sum [\rho A v]_{inlet}$$

As the volume is elemental;

$$\iiint_{cv} \frac{\partial \rho}{\partial t} dU = \frac{\partial \rho}{\partial t} \Delta x \Delta y \Delta z$$

$$\text{Again, } \sum (\rho A v)_{outlet} = [\rho u + \frac{\partial(\rho u)}{\partial x} \Delta x] \Delta y \Delta z + [\rho v + \frac{\partial(\rho v)}{\partial y} \Delta y] \Delta x \Delta z + [\rho w + \frac{\partial(\rho w)}{\partial z} \Delta z] \Delta x \Delta y$$

Therefore, the RTT becomes,

$$0 = \frac{\partial \rho}{\partial t} \Delta x \Delta y \Delta z + \frac{\partial(\rho u)}{\partial x} \Delta x \Delta y \Delta z + \frac{\partial(\rho v)}{\partial y} \Delta x \Delta y \Delta z + \frac{\partial(\rho w)}{\partial z} \Delta x \Delta y \Delta z$$

As the volume is arbitrary & non-zero quantity,

$$\frac{\partial \rho}{\partial t} + \frac{\partial(\rho u)}{\partial x} + \frac{\partial(\rho v)}{\partial y} + \frac{\partial(\rho w)}{\partial z} = 0$$

This is the **Differential Equation of Mass Conservation**.