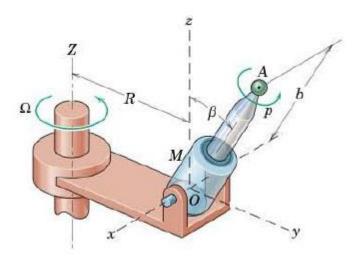
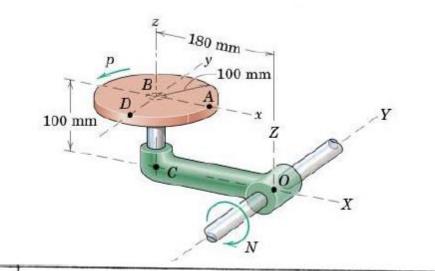
Ques.1 – The small motor *M* is pivoted about the *x*-axis through *O* and gives its shaft *OA* a constant speed *p* rad/s in the direction shown relative to its housing. The entire unit is then set into rotation about the vertical *Z*-axis at the constant angular velocity Ω rad/s. simultaneously, the motor pivots about the x-axis at the constant rate $\dot{\beta}$ for an interval of motion. Determine the angular acceleration α of the shaft *OA* in term of β . Express your result in term of the unit vector for the rotating *x*-*y*-*z* axes.

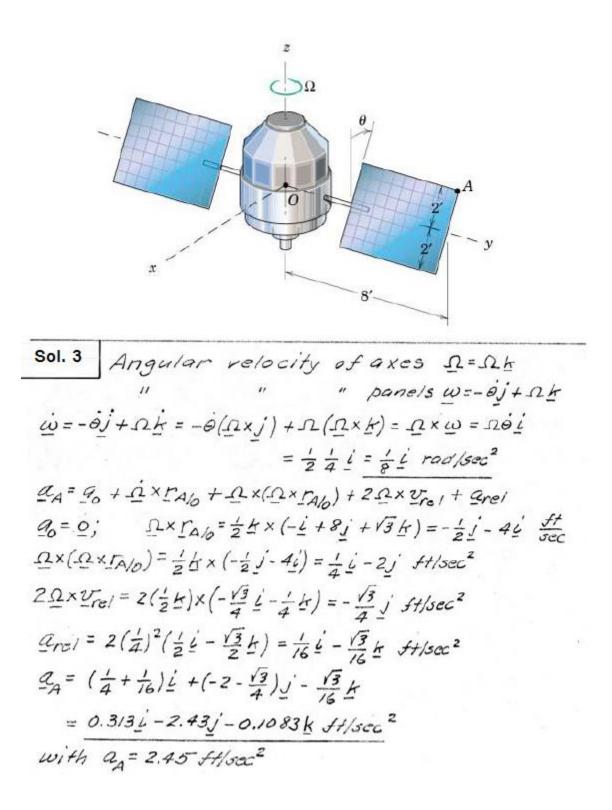


SOL. 1 Angular velocity of OA is $\omega = -\dot{\beta}\underline{i} + \beta \sin\beta\underline{j} + (\beta \cos\beta + \Omega)\underline{k}$ Eq. 7/7a, $[] = \omega$, $(\frac{d[]}{dt})_{XYE} = (\frac{d[]}{dt})_{XYE} + \Omega \times []$ $(\frac{d\omega}{dt})_{XYE} = \Omega + p\dot{\beta}\cos\beta\underline{j} + (-p\dot{\beta}\sin\beta + 0)\underline{k}$ $\underline{\Omega} \times \underline{\omega} = \Omega \cdot \underline{k} \times (-\dot{\beta}\underline{i} + \beta \sin\beta\underline{j} + [p\cos\beta + \Omega]\underline{k})$ $= -\Omega \cdot \underline{\beta}\underline{j} - \Omega \cdot p\sin\beta\underline{i}$ So $\underline{\alpha} = (p\dot{\beta}\cos\beta - \Omega \cdot \underline{\beta})\underline{j} - \Omega \cdot p\sin\beta\underline{i} - p\beta\sin\beta\underline{k}$ $\underline{\alpha} = -\Omega \cdot p\sin\beta\underline{i} + \dot{\beta}(p\cos\beta - \Omega)\underline{j} - p\dot{\beta}\sin\beta\underline{k}$

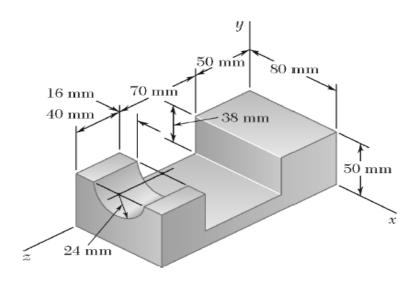
Ques.2 – The circular disk of 100-mm radius rotates about its *z*-axis at the constant speed p = 240 rev/min, and arm *OCB* rotates about *Y*-axis at the constant speed N = 30 rev/min. Determine the velocity **v** and acceleration **a** of point *A* on the disk as it passes the position shown. Use reference axes *x*-*y*-*z* attached to arm *OCB*.



2. SOL $\begin{array}{c|c}
\Omega = angular \quad velocity \quad of \quad axes \quad x-y-z = \frac{2\pi N}{60} j = \pi j \frac{\pi a}{5} \\
\overline{U} = U_{A} = U_{B} + \Omega \times \Gamma_{A/B} + U_{rel} \\
where \quad U_{B} = \pi j \times \Gamma_{0B} = \pi j \times (-0.18\underline{i} + 0.1\underline{k}) = \pi (0.1\underline{i} + 0.18\underline{k}) \text{ m/s} \\
\Omega \times \Gamma_{A/B} = \pi j \times 0.1\underline{i} = -0.1\pi\underline{k} \quad m/s \\
Vrel = P\underline{k} \times \Gamma_{A/B} = \frac{240(2\pi)}{60}\underline{k} \times 0.1\underline{i} = 0.8\pi\underline{j} \quad m/s \\
Collect ferms \quad & get \quad \underline{U} = \pi (0.1\underline{i} + 0.8\underline{j} + 0.08\underline{k}) \quad m/s \\
\Omega = \Omega_{A} = \Omega_{B} + \Omega \times \Gamma_{A/B} + \Omega \times (\Omega \times \Gamma_{A/B}) + 2\Omega \times V_{rel} + \alpha_{rel} \quad \Omega = 0 \\
where \quad \Omega_{B} = \Omega \times (\Omega \times \Gamma_{A/B} + \Omega \times (\Omega \times \Gamma_{A/B}) + 2\Omega \times V_{rel} + \alpha_{rel} \quad \Omega = 0 \\
= \pi^{2} (0.18\underline{i} - 0.1\underline{k}) \quad m/s^{2} \\
\Omega \times (\Omega \times \Gamma_{A/B}) = \pi j \times (-0.1\pi\underline{k}) = -0.1\pi^{2}\underline{i} \quad m/s^{2} \\
\Omega \times (\Omega \times \Gamma_{A/B}) = \pi j \times (0.1\pi\underline{k}) = -0.1\pi^{2}\underline{i} \quad m/s^{2} \\
\Omega = \Omega_{rel} = \dot{p}\underline{k} \times \Gamma_{A/B} + \Gamma_{A/B} \mu^{2}(-\underline{i}) = 0 \\
Qrel = \dot{p}\underline{k} \times \Gamma_{A/B} + \Gamma_{A/B} \mu^{2}(-\underline{i}) = 0 \\
Qrel = \dot{p}\underline{k} \times \Gamma_{A/B} + \Gamma_{A/B} \mu^{2}(-\underline{i}) = 0 \\
\Omega = -0.1\pi^{2}\underline{i} - 6.4\pi^{2}\underline{i} + 0.18\pi^{2}\underline{i} - 0.1\pi^{2}\underline{k} \\
\Omega = -\pi^{2} (6.32\underline{i} + 0.1\underline{k}) \quad m/s^{2}
\end{array}$ Ques.3- The center O of the spacecraft is moving through space with a constant velocity. During the period of motion prior to stabilization, the spacecraft has a constant rotational rate $\Omega = \frac{1}{2}$ rad/sec about its *z*-axis. The x-y-z axes are attached to the body of the craft, and the solar panels rotate about the y-axis at the constant rate $\dot{\theta} = \frac{1}{4}$ rad/sec with respect to the spacecraft. If ω is the absolute angular velocity of the solar panels, determine $\dot{\omega}$. Also find the acceleration of point A when $\theta = 30^{\circ}$.



Ques.4– Determine the mass moment of inertia of the steel fixture shown with respect to (*a*) the *x*-axis, (b) the y-axis, (c) the z-axis. (The density of steel is 7850 kg/m^3 .)



SOLUTION

First compute the mass of each component. We have

Then

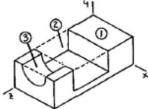
$$m = \rho_{\rm ST} V$$

$$m_1 = 7850 \text{ kg/m}^3 \times (0.08 \times 0.05 \times 0.160) \text{ m}^3$$

$$= 5.02400 \text{ kg}$$

$$m_2 = 7850 \text{ kg/m}^3 \times (0.08 \times 0.038 \times 0.07) \text{ m}^3 = 1.67048 \text{ kg}$$

$$m_3 = 7850 \text{ kg/m}^3 \times \left(\frac{\pi}{2} \times 0.024^2 \times 0.04\right) \text{ m}^3 = 0.28410 \text{ kg}$$



Using Figure 9.28 for components 1 and 2 and the equations derived above for component 3, we have

(a)
$$I_{x} = (I_{x})_{1} - (I_{x})_{2} - (I_{x})_{3}$$

$$= \left\{ \frac{1}{12} (5.02400 \text{ kg}) [(0.05)^{2} + (0.16)^{2}] \text{ m}^{2} + (5.02400 \text{ kg}) \left[\left(\frac{0.05}{2} \right)^{2} + \left(\frac{0.16}{2} \right)^{2} \right] \text{ m}^{2} \right\}$$

$$- \left\{ \frac{1}{12} (1.67048 \text{ kg}) [(0.038)^{2} + (0.07)^{2}] \text{ m}^{2} + (1.67048 \text{ kg}) \left[\left(0.05 - \frac{0.038}{2} \right)^{2} + \left(0.05 + \frac{0.07}{2} \right)^{2} \right] \text{ m}^{2} \right\}$$

$$- \left\{ (0.28410 \text{ kg}) \left[\left(\frac{1}{4} - \frac{16}{9\pi^{2}} \right) (0.024)^{2} + \frac{1}{12} (0.04)^{2} \right] \text{ m}^{2} \right\}$$

$$+ (0.28410 \text{ kg}) \left[\left(0.05 - \frac{4 \times 0.024}{3\pi} \right)^{2} + \left(0.16 - \frac{0.04}{2} \right)^{2} \right] \text{ m}^{2} \right\}$$

$$= [(11.7645 + 35.2936) - (0.8831 + 13.6745) - (0.0493 + 6.0187)] \times 10^{-3} \text{ kg} \cdot \text{ m}^{2}$$

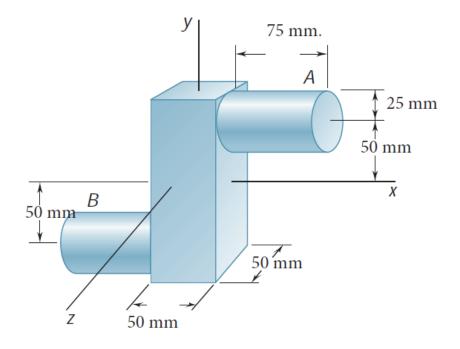
$$= (47.0581 - 14.5576 - 6.0680) \times 10^{-3} \text{ kg} \cdot \text{ m}^{2}$$

$$= 26.4325 \times 10^{-3} \text{ kg} \cdot \text{ m}^{2}$$

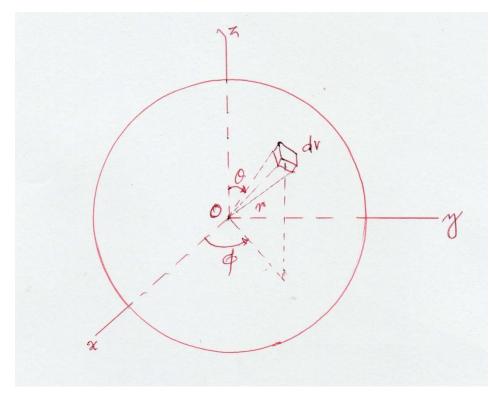
$$\begin{aligned} (b) \quad I_{y} &= (I_{y})_{1} - (I_{y})_{2} - (I_{y})_{3} \\ &= \left\{ \frac{1}{12} (5.02400 \text{ kg}) [(0.08)^{2} + (0.16)^{2}] \text{ m}^{2} + (5.02400 \text{ kg}) \left[\left(\frac{0.08}{2} \right)^{2} + \left(\frac{0.16}{2} \right)^{2} \right] \text{ m}^{2} \right\} \\ &- \left\{ \frac{1}{12} (1.67048 \text{ kg}) [(0.08)^{2} + (0.07)^{2}] \text{ m}^{2} + (1.67048 \text{ kg}) \left[\left(\frac{0.08}{2} \right)^{2} + \left(0.05 + \frac{0.07}{2} \right)^{2} \right] \text{ m}^{2} \right\} \\ &- \left\{ \frac{1}{12} (0.28410 \text{ kg}) [3(0.024)^{2} + (0.04)^{2}] \text{ m}^{2} + (0.28410 \text{ kg}) \left[\left(\frac{0.08}{2} \right)^{2} + \left(0.16 - \frac{0.04}{2} \right)^{2} \right] \text{ m}^{2} \right\} \\ &= [(13.3973 + 40.1920) - (1.5730 + 14.7420) - (0.0788 + 6.0229)] \times 10^{-3} \text{ kg} \cdot \text{m}^{2} \\ &= (53.5893 - 16.3150 - 6.1017) \times 10^{-3} \text{ kg} \cdot \text{m}^{2} \\ &= 31.1726 \times 10^{-3} \text{ kg} \cdot \text{m}^{2} \end{aligned}$$

$$\begin{aligned} (c) & I_z = (I_z)_1 - (I_z)_2 - (I_z)_3 \\ &= \left\{ \frac{1}{12} (5.02400 \text{ kg}) [(0.08)^2 + (0.05)^2] \text{ m}^2 + (5.02400 \text{ kg}) \left[\left(\frac{0.08}{2} \right)^2 + \left(\frac{0.05}{2} \right)^2 \right] \text{ m}^2 \right\} \\ &- \left\{ \frac{1}{12} (1.67048 \text{ kg}) [(0.08)^2 + (0.038)^2] \text{ m}^2 + (1.67048 \text{ kg}) \left[\left(\frac{0.08}{2} \right)^2 + \left(0.05 - \frac{0.038}{2} \right)^2 \right] \text{ m}^2 \right\} \\ &- \left\{ (0.28410 \text{ kg}) \left(\frac{1}{2} - \frac{16}{9\pi^2} \right) (0.024 \text{ m})^2 + (0.28410 \text{ kg}) \left[\left(\frac{0.08}{2} \right)^2 + \left(0.05 - \frac{4 \times 0.024}{3\pi} \right)^2 \right] \text{ m}^2 \right\} \\ &= [(3.7261 + 11.1784) - (1.0919 + 4.2781) - (0.0523 + 0.9049)] \times 10^{-3} \text{ kg} \cdot \text{m}^2 \\ &= (14.9045 - 5.3700 - 0.9572) \times 10^{-3} \text{ kg} \cdot \text{m}^2 \\ &= 8.5773 \times 10^{-3} \text{ kg} \cdot \text{m}^2 \end{aligned}$$

Ques.5 – A steel forging consists of a 150 x 50 x 50-mm rectangular prism and two cylinders of diameter 50 mm and length 75 mm as shown. Determine the moments of inertia of the forging with respect to the coordinate axes, knowing that the specific weight of steel is 7850 kg/m³.



Ques.6 Compute the component of inertia tensor at the center of a solid sphere of uniform density ρ as shown in Fig.



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With the aid of integration formulas, we have

$$Iyy = \int_{0}^{R} \int_{0}^{2\pi} \pi^{4} \cos^{2}\phi \left[-\frac{1}{3} \cos \left((kn^{2} + \lambda) \right] \right]^{\pi} d\phi dn + \int_{0}^{R} \int_{0}^{2\pi} \pi^{4} \left(-\frac{\cos^{3}\theta}{3} \right) \int_{0}^{\pi} d\phi dn = \int_{0}^{R} \int_{0}^{2\pi} \pi^{4} \cos^{2}\phi \cdot \frac{1}{3} d\phi dn + \int_{0}^{R} \int_{0}^{\pi} (r^{4}) \frac{1}{3} d\phi dn$$
Integrating next with respect to ϕ we get

$$Iyy = \int_{0}^{R} (n^{4}) (\frac{1}{3}) (\pi) dn + \int_{0}^{R} (r^{4}) (\frac{2}{3}) (2\pi) dn$$
Finally we get

$$Iyy = \int_{0}^{R} \frac{\pi^{5}}{3} \frac{4}{3} \pi + \int_{0}^{R} \frac{\pi^{5}}{3} \frac{4}{3} \pi$$

$$\therefore Iyy = \int_{0}^{R} \frac{\pi^{5}}{3} \frac{4}{3} \pi + \int_{0}^{R} \frac{\pi^{5}}{3} \frac{4}{3} \pi$$

$$\therefore Iyy = \int_{0}^{R} \frac{\pi^{7}}{3} \frac{4}{3} \pi R^{5}$$
But $M = \int_{0}^{\frac{4}{3}} \pi R^{3}$, Hence $Iyy = \frac{3}{5} MR^{2}$
Because of the point symmetry about point 0, we can

oulso say that

$$7ar = I_{rr} = \frac{2}{r} MR^2$$

Because the coordonate planes are all planes of symmetry for the mass destribution, the product of inertia are zero. Thus the inertia tensor can be given as

$$I_{ij} = \begin{bmatrix} \frac{2}{5}MR^2 & 0 & 0 \\ 0 & \frac{2}{5}MR^2 & 0 \\ 0 & 0 & \frac{2}{5}MR^2 \end{bmatrix}$$