## Indian Institute of Technology Guwahati

ME 101: Engineering Mechanics (2016-2017, Sem II) Quiz-2 Solution(3.04.2017)

Time: 8:00 AM – 8:45 AM Full Marks: 60

Q1. The acceleration of a particle in cylindrical coordinate system  $(r, \theta, z)$  is given by

$$\vec{a} = (\ddot{r} - r \dot{\theta}^2) \hat{e}_r + (r \ddot{\theta} + 2 \dot{r} \dot{\theta}) \hat{e}_{\theta} + \ddot{z} \hat{e}_z$$

where  $\hat{e}_r$ ,  $\hat{e}_\theta$  and  $\hat{e}_z$  are, respectively, the unit vectors in the instantaneous radial, azimuthal and axial directions of the particle. The rate of change of acceleration  $\left(\frac{d\vec{a}}{dt}\right)$  is called jerk.

- (a) Derive an expression for the jerk of a particle in cylindrical polar coordinates.
- (b) Calculate the jerk experienced by a small insect crawling with a constant speed of **1 cm/s** on a helical spring of **2 cm** diameter and 1 cm pitch. [15+5=20 marks]

## **Solution Q1-**

(a)

$$\begin{split} \vec{a} &= \left(\ddot{r} - r \, \dot{\theta}^2\right) \, \hat{e}_r + \left(r \, \ddot{\theta} + \, 2 \, \dot{r} \, \dot{\theta}\right) \, \hat{e}_\theta + \ddot{z} \, \hat{e}_z \\ \frac{d\vec{a}}{dt} &= \frac{d}{dt} \left[ \left(\ddot{r} - r \, \dot{\theta}^2\right) \, \hat{e}_r \right] + \frac{d}{dt} \left[ \left(r \, \ddot{\theta} + \, 2 \, \dot{r} \, \dot{\theta}\right) \, \hat{e}_\theta \right] + \frac{d}{dt} \left[ \ddot{z} \, \hat{e}_z \right] \\ &= \frac{d}{dt} \left(\ddot{r} - r \, \dot{\theta}^2\right) \, \hat{e}_r + \left(\ddot{r} - r \, \dot{\theta}^2\right) \frac{d\hat{e}_r}{dt} \\ &+ \frac{d}{dt} \left(r \, \ddot{\theta} + \, 2 \, \dot{r} \, \dot{\theta}\right) \hat{e}_\theta + \left(r \, \ddot{\theta} + \, 2 \, \dot{r} \, \dot{\theta}\right) \frac{d\hat{e}_\theta}{dt} \\ &+ \frac{d\ddot{z}}{dt} \hat{e}_z + \ddot{z} \frac{d\hat{e}_z}{dt} \\ &\frac{d}{dt} \left(\ddot{r} - r \, \dot{\theta}^2\right) = \ddot{r} - \dot{r} \dot{\theta}^2 - 2 \, r \, \dot{\theta} \, \ddot{\theta} \\ &\frac{d}{dt} \left(r \, \ddot{\theta} + \, 2 \, \dot{r} \, \dot{\theta}\right) = \dot{r} \ddot{\theta} - r \, \ddot{\theta} + 2 \, \dot{r} \, \dot{\theta} + 2 \, \dot{r} \, \ddot{\theta} \\ &\left|\frac{d\hat{e}_r}{d\theta}\right| = \lim_{\Delta\theta \to 0} \left|\frac{\Delta\hat{e}_r}{\Delta\theta}\right| = \lim_{\Delta\theta \to 0} \left|\frac{2 \sin \frac{\Delta\theta}{2}}{\Delta\theta}\right| = \lim_{\Delta\theta \to 0} \left|\frac{\sin \frac{\Delta\theta}{2}}{\Delta\theta}\right| = 1 \\ &\left|\frac{d\hat{e}_\theta}{d\theta}\right| = \lim_{\Delta\theta \to 0} \left|\frac{\Delta\hat{e}_\theta}{\Delta\theta}\right| = \lim_{\Delta\theta \to 0} \left|\frac{2 \sin \frac{\Delta\theta}{2}}{\Delta\theta}\right| = \lim_{\Delta\theta \to 0} \left|\frac{\sin \frac{\Delta\theta}{2}}{\Delta\theta}\right| = 1 \\ &\left|\frac{d\hat{e}_\theta}{d\theta}\right| = \lim_{\Delta\theta \to 0} \left|\frac{\Delta\hat{e}_\theta}{\Delta\theta}\right| = \lim_{\Delta\theta \to 0} \left|\frac{2 \sin \frac{\Delta\theta}{2}}{\Delta\theta}\right| = \lim_{\Delta\theta \to 0} \left|\frac{\sin \frac{\Delta\theta}{2}}{\Delta\theta}\right| = 1 \end{aligned}$$

The directions of  $\frac{d\hat{e}_r}{d\theta}$  and  $\frac{d\hat{e}_{\theta}}{d\theta}$  are along increasing  $\theta$  and decreasing r, respectively. Therefore,

$$\frac{d\hat{e}_r}{d\theta} = \hat{e}_\theta$$

$$\frac{d\hat{e}_{\theta}}{d\theta} = -\hat{e}_r$$

$$\begin{split} \frac{d\hat{e}_r}{dt} &= \frac{d\hat{e}_r}{d\theta} \frac{d\theta}{dt} = \dot{\theta} \frac{d\hat{e}_r}{d\theta} = \dot{\theta} \; \hat{e}_{\theta} \\ \frac{d\hat{e}_{\theta}}{dt} &= \frac{d\hat{e}_{\theta}}{d\theta} \frac{d\theta}{dt} = \dot{\theta} \frac{d\hat{e}_{\theta}}{d\theta} = -\dot{\theta} \; \hat{e}_r \end{split}$$

$$\begin{split} \frac{d\vec{a}}{dt} &= \frac{d}{dt} \left[ \left( \ddot{r} - r \, \dot{\theta}^{\, 2} \right) \, \hat{e}_{r} \right] + \frac{d}{dt} \left[ \left( r \, \ddot{\theta} + \, 2 \, \dot{r} \, \dot{\theta} \right) \, \hat{e}_{\theta} \right] + \frac{d}{dt} \left[ \ddot{z} \, \hat{e}_{z} \right] \\ &= \frac{d}{dt} \left( \ddot{r} - r \, \dot{\theta}^{\, 2} \right) \, \hat{e}_{r} + \left( \ddot{r} - r \, \dot{\theta}^{\, 2} \right) \frac{d\hat{e}_{r}}{dt} \\ &+ \frac{d}{dt} \left( r \, \ddot{\theta} + \, 2 \, \dot{r} \, \dot{\theta} \right) \hat{e}_{\theta} + \left( r \, \ddot{\theta} + \, 2 \, \dot{r} \, \dot{\theta} \right) \frac{d\hat{e}_{\theta}}{dt} \\ &+ \frac{d\ddot{z}}{dt} \, \hat{e}_{z} + \ddot{z} \frac{d\hat{e}_{z}}{dt} \\ &\frac{d\ddot{a}}{dt} = \left( \ddot{r} - \dot{r} \, \dot{\theta}^{\, 2} - 2 \, r \, \dot{\theta} \, \ddot{\theta} \right) \, \hat{e}_{r} + \left( \ddot{r} - r \, \dot{\theta}^{\, 2} \right) \, \dot{\theta} \, \hat{e}_{\theta} \\ &+ \left( \dot{r} \ddot{\theta} + r \, \ddot{\theta} + 2 \, \ddot{r} \, \dot{\theta} + 2 \, \dot{r} \, \ddot{\theta} \right) \, \hat{e}_{\theta} + \left( r \, \ddot{\theta} + \, 2 \, \dot{r} \, \dot{\theta} \right) \left( -\dot{\theta} \, \hat{e}_{r} \right) \\ &+ \ddot{z} \, \hat{e}_{z} + 0 \end{split}$$

Collecting components along  $\hat{e}_r$ ,  $\hat{e}_{ heta}$  and  $\hat{e}_z$  directions,

$$\begin{split} \frac{d\vec{a}}{dt} &= \left(\ddot{r} - \dot{r}\,\dot{\theta}^2 - 2\,r\,\dot{\theta}\,\ddot{\theta} - r\,\dot{\theta}\,\ddot{\theta} - 2\,\dot{r}\,\dot{\theta}^2\right)\hat{e}_r \\ &\quad + \left(\ddot{r}\,\dot{\theta} - r\,\dot{\theta}^3 + \dot{r}\,\ddot{\theta} + r\,\ddot{\theta} + 2\,\ddot{r}\,\dot{\theta} + 2\,\dot{r}\,\ddot{\theta}\right)\hat{e}_\theta + \ddot{z}\,\hat{e}_z \\ \\ \frac{d\vec{a}}{dt} &= \left(\ddot{r} - 3\,\dot{r}\,\dot{\theta}^2 - 3\,r\,\dot{\theta}\,\ddot{\theta}\right)\hat{e}_r + \left(3\,\ddot{r}\,\dot{\theta} + 3\,\dot{r}\,\ddot{\theta} - r\,\dot{\theta}^3 + r\,\ddot{\theta}\right)\hat{e}_\theta + \ddot{z}\,\hat{e}_z \end{split}$$

(b)

r = R = 0.01 m = const.,

$$\alpha = \tan^{-1} \left( \frac{P}{2\pi R} \right) = \tan^{-1} \left( \frac{0.01}{2\pi \times 0.01} \right) = 0.15915$$

$$\dot{\theta} = \frac{v_{\theta}}{R} = \frac{v \cos \alpha}{R} = 0.987 \text{ s}^{-1} = \text{const.},$$

$$\dot{z} = v_z = v \sin \alpha = \text{const.}$$

$$\frac{d\vec{a}}{dt} = (0 - 3 \times 0 \times \dot{\theta}^2 - 3R \times \dot{\theta} \times 0) \,\hat{e}_r + (3 \times 0 \times \dot{\theta} + 3 \times 0 \times 0 - R \,\dot{\theta}^3 + R \times 0) \,\hat{e}_\theta + 0 \,\hat{e}_z$$

$$= 0 \,\hat{e}_r + (-R \,\dot{\theta}^3) \,\hat{e}_\theta + 0 \,\hat{e}_z$$

$$= -R \,\dot{\theta}^3 \,\hat{e}_\theta$$

$$= -0.01 \times (0.987)^3 \,\hat{e}_\theta$$

$$= -0.00961 \,\hat{e}_\theta \,\text{m/s}^3$$

Q2. At the instant shown in **Fig.1**, the rod **AB** is rotating about the **z** axis with an angular velocity  $\omega_1 = 4 \, rad/s$  and an angular acceleration  $\dot{\omega}_1 = 3 \, rad/s^2$ . At this same instant, the circular rod has an angular motion relative to the rod as shown. If the collar *C* is moving down around the circular rod with a speed of **7.6** cm/s, increasing at a rate **20**.  $3 \, cm/s^2$ , both measured relative to the rod.

Determine (a) the collar's velocity and, (b) acceleration at this instant. [5+15=20 marks]

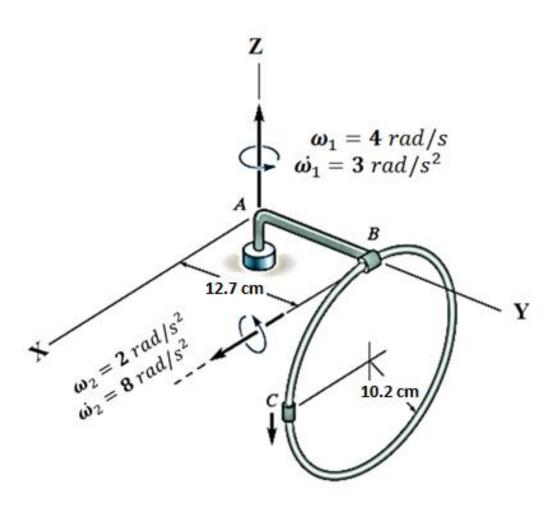


Fig. 1

## SOLUTION:

$$\begin{split} &\Omega = \{4 \, k\} \, rad/s \\ &\dot{\Omega} = \{3 \, k\} \, rad/s^2 \\ &\mathbf{r}_B = \{0.127 \, j\} \, m \\ &\mathbf{v}_B = (4 \, k) \, \times \, (0.127 \, j) = \{-0.508 \, i\} \, m/s \\ &\mathbf{a}_B = \, \ddot{\mathbf{r}}_B = \left[ (\ddot{\mathbf{r}}_B)_{xyz} + \, \Omega \, \times (\dot{\mathbf{r}}_B)_{xyz} \, \right] + \, \dot{\Omega} \, \times \mathbf{r}_B + \, \Omega \, \times \, \dot{\mathbf{r}}_B \\ &= (4 \, k) \, \times (-0.508 \, i) \, + \, (3 \, k) \, \times (0.127 \, j) \\ &= \{-0.381 \, i - 2.032 \, j\} \, m/s^2 \end{split}$$

$$\begin{split} &\Omega_{C/B} = \{2 \, \mathbf{i}\} \, \mathrm{rad/s} \\ &\dot{\Omega}_{C/B} = \{8 \, \mathbf{i}\} \, \mathrm{rad/s^2} \\ &\mathbf{r}_{C/B} = \{0.1016 \, \mathbf{i} - 0.1016 \, \mathbf{k}\} \, \mathrm{m} \\ &\left(\mathbf{v}_{C/B}\right)_{xyz} = \left(\dot{\mathbf{r}}_{C/B}\right) \, + \, \Omega_{C/B} \times \mathbf{r}_{C/B} \\ &= \left(-0.0762 \, \mathbf{k}\right) \, + \, \left(2 \, \mathbf{i}\right) \times \left(0.1016 \, \mathbf{i} - 0.1016 \, \mathbf{k}\right) \end{split}$$

 $= \{0.2032 \, \mathbf{j} - 0.0762 \, \mathbf{k} \} \, \text{m/s}$ 

$$\begin{split} \left(\mathbf{a}_{\text{C/B}}\right)_{\text{xyz}} &= \left[ \left(\ddot{\mathbf{r}}_{\text{C/B}}\right)_{\text{xyz}} + \, \Omega_{\text{C/B}} \times \left(\dot{\mathbf{r}}_{\text{C/B}}\right)_{\text{xyz}} \right] + \, \dot{\Omega}_{\text{C/B}} \times \mathbf{r}_{\text{C/B}} + \, \Omega_{\text{C/B}} \times \dot{\mathbf{r}}_{\text{C/B}} \\ &= \left[ \left( \frac{0.0762^2}{0.1016} \, \mathbf{i} \, - 0.2032 \, \mathbf{k} \right) + \, (2 \, \mathbf{i}) \times (-0.0762 \, \mathbf{k}) \right] + \, (8 \, \mathbf{i}) \\ &\times (0.1016 \, \mathbf{i} - 0.1016 \, \mathbf{k}) + \, (2 \, \mathbf{i}) \times (0.2032 \, \mathbf{j} - 0.0762 \, \mathbf{k}) \end{split}$$

=  $\{-0.05715 \, \mathbf{i} + 1.1176 \, \mathbf{j} + 0.2032 \, \mathbf{k}\} \, \text{m/s}^2$ 

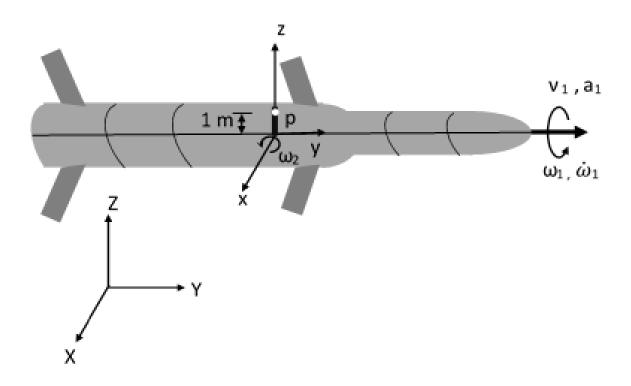
$$\begin{aligned} \mathbf{v}_{C} &= \mathbf{v}_{B} + \Omega \times \mathbf{r}_{C/B} + (\mathbf{v}_{C/B})_{xyz} \\ &= (-0.508 \, \mathbf{i}) + (4 \, \mathbf{k}) \times (0.1016 \, \mathbf{i} - 0.1016 \, \mathbf{k}) + (0.2032 \, \mathbf{j} - 0.0762 \, \mathbf{k}) \\ &= \{-0.508 \, \mathbf{i} + 0.6096 \, \mathbf{j} - 0.0762 \, \mathbf{k}\} \, \text{m/s} \\ \mathbf{a} &= \mathbf{a}_{B} + \dot{\Omega} \times \mathbf{r}_{C/B} + \Omega \times (\Omega \times \mathbf{r}_{C/B}) + 2\Omega \times (\mathbf{v}_{C/B})_{xyz} + (\mathbf{a}_{C/B})_{xyz} \end{aligned}$$

$$= (-0.381 \,\mathbf{i} - 2.032 \,\mathbf{j}) + (3 \,\mathbf{k}) \times (0.1016 \,\mathbf{i} - 0.1016 \,\mathbf{k}) + (4 \,\mathbf{k}) \\ \times [(4 \,\mathbf{k}) \times (0.1016 \,\mathbf{i} - 0.1016 \,\mathbf{k})] + 2(4 \,\mathbf{k}) \times (0.2032 \,\mathbf{j} - 0.0762 \,\mathbf{k}) \\ + (-0.05715 \,\mathbf{i} + 1.1176 \,\mathbf{j} + 0.2032 \,\mathbf{k})$$

$$= \{-3.683 \,\mathbf{i} - 0.6096 \,\mathbf{j} + 0.2032 \,\mathbf{k}\} \,\mathbf{m/s^2}$$

- Q3. A missile travels in a straight line with respect to inertial reference XYZ at speed  $v_1 = 4$  m/s and change of speed  $a_1 = 2$  m/s<sup>2</sup> as shown in Fig. 2. At the same instant, the missile rolls about its direction of flight at an angular speed  $\omega_1 = 5$  rad/s and change in angular speed  $\dot{\omega}_1 = 5$  rad/s<sup>2</sup> as shown in the Fig. 2. Inside the missile a rod is rotating at a constant angular speed (relative to the missile)  $\omega_2 = 10$  rad/s about an axis perpendicular to the page. A point p on the rod moves outward with speed 1 m/s with respect to the rod. The point p on the rod is at a distance of 1m from the axis of rotation. Determine the following
- (a) Find the angular velocity and angular acceleration of rod with respect to ground (XYZ).
- (b) Find Coriolis acceleration  $(2\vec{\omega} \times \vec{V})$  of Point p in the frame fixed with the rod and the frame fixed with the missile. [15+5=20 marks]

Fig 2.



angular velocity of 
$$\omega_2 = \omega_1 j = 5j$$
 radis angular accord  $\omega_2 = \omega_1 j = 5j$  radis  $\omega_1 j = 5j$  radis  $\omega_2 = 0$ 

angular velocity of rot  $\omega_3 = \omega_2 \dot{c}$  rad/r  $\omega_3 = \omega_2 \dot{c}$  rad/r  $\omega_3 = 0$  rad/s angular acceleration of  $\omega_3 = 0$  rad/s  $\omega_3 = 0$ 

velocity of missite 19/1 = 0,1 = 4jm/s

accoladir of missite. 92/1 = aij 2jm/s ~

Dyder velocity of of rodo co. ril gran

$$\frac{\omega_3}{||_1 = \omega_3|_2 + \omega_2|_1 - 0}$$

$$\frac{||\omega_3||_1 = ||0|\underline{i} + ||5||_1 + ||5||_1}{||\omega_3||_1 = ||0|\underline{i} + ||5||_1}$$

arguler acceleration of rod wint grow  $|\dot{\omega}_3| = |\dot{\omega}_{3/2}| + |\dot{\omega}_{21}|$ 

$$= \frac{\omega_3|_2 + \omega_2|_{x} \omega_3|_2 + \omega_2|_1}{= 0 + 5j \times 10i + 50j}$$

$$= \frac{\omega_3|_2 + \omega_2|_{x} \times \omega_3|_2 + \omega_2|_1}{|\omega_3|_2 - |\omega_3|_2 - |\omega_3|_2}$$

velocity of part p ion with respect to rod υρ/8 = 1 k m/s angular velocity of rod w. r.t grow W3/1 = 101+51 corjohis acceleration in frame attach to rod acoi = 2 wx o = 2 03/X @ Op/3 =2(10i+51) X14 aco = -20 j+ 10 c/ m/s2 2p/2 = 1k + 63x x

volouty of pant pin frame attached to

consus acceleration of pin frame allache

$$\frac{Q_{co} = 2 \frac{\omega_{2}}{|X|} \frac{|X|}{|Y|} \frac{|Q|}{|X|}$$

$$= 2 \times \frac{5}{|X|} \frac{|X|}{|Y|} \frac{|Y|}{|Y|} \frac{|Y|}{|Y|}$$

$$= \frac{10 \dot{C}}{|Y|} - \frac{10 \dot{C}}{|Y|} \frac{|Y|}{|Y|} \frac{|X|}{|X|}$$